

SUPPLEMENTAL TYPE CERTIFICATE

10051957

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AIR ALLIANCE GmbH

**FLUGHAFEN SIEGERLAND
WERFTHALLE G1-G3
57299 BURBACH
GERMANY**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Type Certificate Number : US A10CE
Type Certificate Holder : LEARJET INC.
Type Design - Model : 35, 35A**

Description of Design Change:

Installation of Dual Garmin GTN 750

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

Special Condition(s): F-04, HIRF Protection.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 21 January 2015



**Alain LEROY
Head of Large Aeroplanes Department**

Note:
The following numbers are listed on the certificate:
EASA current Project Number: 0010029531-001

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Associated Technical Documentation:

Air Alliance, Master Document List, 2014_001_LJ35_Form2, revision 04, dated 16 January 2015.

Air Alliance, Aircraft Flight Manual Supplement (GPS/NAV/COM), 2014_001_LJ35_0506, revision 01, dated 04 November 2014.

Air Alliance, Aircraft Flight Manual Supplement (TAWS A), 2014_001_LJ35_0506a, revision 00, dated 22 September 2014.

Air Alliance, Instructions for Continued Airworthiness, 2014_001_LJ35_0507, revision 00, dated 27 August 2014.

or later revisions of the above listed documents approved by EASA.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- end -

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